

Consistent time-step optimization in the Lattice Boltzmann method

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- 1 Context
- 2 Physical aspects of the adaptive time-stepping
- 3 Algorithm
- 4 Validation
 - Natural convection in thermal flow
 - Womersley flow
 - Channel entrance flow
- 5 Conclusion

Context

Lattice Boltzmann method is a great tool but

- Due to the low symmetry of standard lattices, standard stream-and-collide LB algorithm reduces to an isothermal weakly compressible Navier-Stokes model:

$$\text{Ma} = \frac{|\mathbf{u}|_{\max}}{c_s} \leq 0.3.$$

- LB method is by nature a compressible method.
- ↪ Extensive research focuses on lifting the restrictions to low Mach numbers and isothermal fluids in LB approach.
- ↪ Incompressible LB models only decrease the order of compressibility errors in steady flows.

Why incompressible flows matter ?

The maximum time-step is, in general, expressed as

$$\Delta t_{\max} = \frac{\text{CFL } \Delta x}{v_{\max}}, \text{ where } v_{\max} = c_s + |\mathbf{u}|_{\max}$$

Courant-Friedrichs-Lewy (CFL) number: normalized maximum velocity at which flow variations can be robustly propagated by numerical scheme.

$$\Delta t_{\max} \approx \frac{3 \times 10^{-6}}{1 + \text{Ma}} \text{ for } \text{CFL} \approx 1, \Delta x \approx 10^{-3} \text{ m}, c_s \approx 343 \text{ m/s}$$

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For a (truly) incompressible fluid

$$\Delta t_{\max} = \frac{\text{CFL} \Delta x}{|\mathbf{u}|_{\max}} \implies \Delta t_{\max}^{\text{incomp.}} \simeq \frac{\Delta t_{\max}^{\text{comp.}}}{\text{Ma}}$$

Artificial speed of sound

- In the LB method, the distribution functions move from one lattice node to another during exactly one time-step, with a characteristic speed $c = \Delta x / \Delta t$.
- On the other hand, the propagation of sound is associated to the effective transport of mass-density variations via the distribution functions.

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↪ Speed of sound and speed of microscopic propagation are physically related, for standard lattice:

$$c = c_0 \sqrt{3} \quad , \quad \text{where } c_0 = \sqrt{p/\rho} \quad \text{is the } \textit{isothermal} \text{ speed of sound}$$

↪ Accelerate a LB simulation is to *artificially* decrease c_0 , or equivalently, to increase the compressibility of the fluid (same as in the artificial compressibility method).

Artificial speed of sound

- × Set a targeted Ma . In unsteady simulations, maximum velocity may vary by orders of magnitude.
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Aims of this presentation

- ⦿ Comment on this impact.
- ⦿ Propose a correction to preserve the continuity of the pressure forces.

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Physical aspects of the adaptive time-stepping

The compressible Navier-Stokes equations with a fluctuated density field $\rho(\mathbf{x}, t) = \rho_{\text{ref}} + \rho'(\mathbf{x}, t)$ read

$$\frac{\partial \rho'}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0 \quad \& \quad \frac{\partial \mathbf{u}}{\partial t} + (\mathbf{u} \cdot \nabla) \mathbf{u} = -\frac{1}{\rho} \nabla (\rho' c_0^2) + \nu \Delta \mathbf{u} + \frac{\mathbf{f}_{\text{ext}}}{\rho}$$

If the speed of sound is changed from c_0 to $c_0^* = \lambda c_0$

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- The continuity of the pressure force per unit mass requires that

$$-\frac{c_0^2}{\rho} \nabla \rho' = -\frac{(\lambda c_0)^2}{\rho^*} \nabla \rho'^* \quad \implies \quad \rho^* = \rho_{\text{ref}} \left(\frac{\rho}{\rho_{\text{ref}}} \right)^{\frac{1}{\lambda^2}}.$$

This yields to a spurious source term in the mass conservation equation:

$$\frac{\partial \rho^{*'}}{\partial t} + \nabla \cdot (\rho^* \mathbf{u}) = \frac{\lambda^2 - 1}{\lambda^2} \rho^* (\nabla \cdot \mathbf{u})$$

$\lambda \simeq 1$ and $\nabla \cdot \mathbf{u} \simeq 0$ in the weakly-compressible regime, this term remains small in practice.

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Subtleties of the algorithm implementation

In the *adaptive time-stepping* algorithm, the speed of sound is tailored in order to maintain a constant target Mach number Ma_t so that

$$c_0^*(t) = \frac{u_{\max}(t)}{\text{Ma}_t} = \lambda(t)c_0(t) \quad \& \quad \Delta t^* = \frac{1}{\lambda} \Delta t.$$

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- The stream-and-collide algorithm is usually solved in non-dimensional lattice units ($\tilde{\square}$) for which $\Delta \tilde{x} = \Delta \tilde{t} = 1$.
- Somewhat against our intuition, the speed of sound in this framework remains constant and equal to $\tilde{c}_0 = 1/\sqrt{3}$.
- The rescaled maximum fluid velocity \tilde{u}_{\max}^* (in lattice units) is adapted *wrt* Ma_t .

Summary of the algorithm

1. Update g_α via streaming by using $g_\alpha(\mathbf{x}, t) = \widehat{g}_\alpha(\mathbf{x} - \mathbf{c}_\alpha \Delta t, t - \Delta t)$.
2. Compute ρ , $\tilde{\mathbf{f}}_{ext}$, $\tilde{\mathbf{u}}$, and g_α^{eq} to obtain g_α^{neq} by using $g_\alpha^{neq} = g_\alpha - g_\alpha^{eq} + \frac{\Delta t}{2} F_\alpha$.
3. Compute \tilde{u}_{\max} and $\lambda = \tilde{u}_{\max} / (\tilde{c}_0 \text{Ma}_t)$.
4. Compute ρ^* for adaptive time-stepping (ATS) with correction.
5. Compute the rescaled variables $\tilde{\mathbf{u}}^* = \tilde{\mathbf{u}} / \lambda$ and $\tilde{\mathbf{f}}_{ext}^* = \tilde{\mathbf{f}}_{ext} / \lambda^2$.
6. Compute g_α^{*eq} and \tilde{F}_α^* with the rescaled variables
7. Compute $\tilde{\tau}_g^*$ by using $\tilde{\tau}_g^* = \frac{1}{\lambda} \left(\tilde{\tau}_g - \frac{1}{2} \right) + \frac{1}{2}$.
8. Compute g_α^{*neq} using $g_\alpha^{*neq} = \frac{1}{\lambda} \frac{\rho^*}{\rho} \frac{\tilde{\tau}_g^*}{\tilde{\tau}_g} g_\alpha^{neq}$ together with g_α^{neq} from step 2.
9. Compute \widehat{g}_α^* by using $\widehat{g}_\alpha^*(\mathbf{x}, t) = g_\alpha^*(\mathbf{x}, t) - \frac{\Delta t^*}{\tau_g^*} g_\alpha^{*neq}(\mathbf{x}, t) + \Delta t^* F_\alpha^*(\mathbf{x}, t)$.

Detailed steps 5. & 7.

$$c_0^* = \lambda c_0 \quad \& \quad \Delta t^* = \frac{1}{\lambda} \Delta t$$

Step 5: rescaled variables \tilde{u}^* , \tilde{f}_{ext}^*

$$\tilde{u}^* = \mathbf{u} \frac{\Delta t^*}{\Delta x} = \mathbf{u} \frac{\Delta t}{\Delta x} \frac{1}{\lambda} = \tilde{u} \frac{1}{\lambda} \quad \& \quad \tilde{f}_{ext}^* = \mathbf{f}_{ext} \frac{(\Delta t^*)^2}{\Delta x} = \frac{1}{\lambda^2} \tilde{f}_{ext}.$$

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Step 7: rescaled relaxation time $\tilde{\tau}_g^*$

Considering that the viscosity must remain unaltered with

$$\nu = \left(\tilde{\tau}_g - \frac{1}{2} \right) \frac{c_0 \Delta x}{\sqrt{3}} = \left(\tilde{\tau}_g^* - \frac{1}{2} \right) \frac{c_0^* \Delta x}{\sqrt{3}} \quad \implies \quad \tilde{\tau}_g^* = \frac{1}{\lambda} \left(\tilde{\tau}_g - \frac{1}{2} \right) + \frac{1}{2}.$$

Detailed step 8. regularization as a possible option

The rescaling of g_α^{neq} is not straightforward since its projection onto moment space includes non-hydrodynamic moments, whose rescaling is not intuitive.

Regularization of g_α^{neq} (as proposed by Latt)

- Rescaling by regularization relies on the continuity of $\sum_\alpha g_\alpha^{neq} \mathbf{c}_\alpha \mathbf{c}_\alpha$.
- This expression does not lead to any particular properties at the macroscopic level.
- Regularization can induce a computational overload.

Detailed step 8.

Alternative rescaling based on the continuity of \mathcal{S}

A Chapman-Enskog analysis establishes that

$$\sum_{\alpha=0}^{q-1} g_{\alpha}^{neq} \mathbf{c}_{\alpha} \mathbf{c}_{\alpha} = -2\rho\tau_g c_0^2 \mathbf{S} + \mathcal{O}(\text{Ma}^3), \quad \text{where } \mathbf{S} \text{ is the rate-of-strain tensor.}$$

The continuity of \mathcal{S} then gives in lattice units

$$\frac{\sum_{\alpha=0}^{q-1} g_{\alpha}^{neq} \mathbf{e}_{\alpha} \mathbf{e}_{\alpha}}{\rho \tilde{\tau}_g \Delta t} = \frac{\sum_{\alpha=0}^{q-1} g_{\alpha}^{*neq} \mathbf{e}_{\alpha} \mathbf{e}_{\alpha}}{\rho^* \tilde{\tau}_g^* \Delta t^*}$$

and eventually yields

$$g_{\alpha}^{*neq} = \frac{1}{\lambda} \frac{\rho^*}{\rho} \frac{\tilde{\tau}_g^*}{\tilde{\tau}_g} g_{\alpha}^{neq}$$

by assuming that all the g_{α}^{neq} 's are rescaled by a same factor.

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Natural convection in thermal flow

Temperature-driven buoyancy force (Boussinesq hypothesis):

$$\mathbf{f}_b(\mathbf{x}, t) = \rho(\mathbf{x}, t) \mathbf{g} \beta (T(\mathbf{x}, t) - T_0)$$

where β is the coefficient of thermal expansion of the fluid, \mathbf{g} is the gravitational acceleration and T_0 is the temperature at rest.

Initial hot spot (plume): $T(\mathbf{x}, t_0) = T_0 + \exp\left(-\frac{x^2+y^2}{R^2}\right) \Delta T$

Hybrid finite-difference scheme / LB scheme

$$\frac{\partial T}{\partial t} + (\mathbf{u} \cdot \nabla) T = \kappa \nabla^2 T \quad \Longrightarrow \quad T(\mathbf{x}, t + 1) = T(\mathbf{x}, t) - (\tilde{\mathbf{u}} \cdot \nabla_h) T + \tilde{\kappa} \Delta_h T$$

where ∇_h and Δ_h stand for finite-difference gradient and Laplacian operators.

$$\tilde{\kappa}^* = \nu \Delta t^* / Pr \Delta x^2 \quad \text{with the Prandtl number } Pr = \nu / \kappa.$$

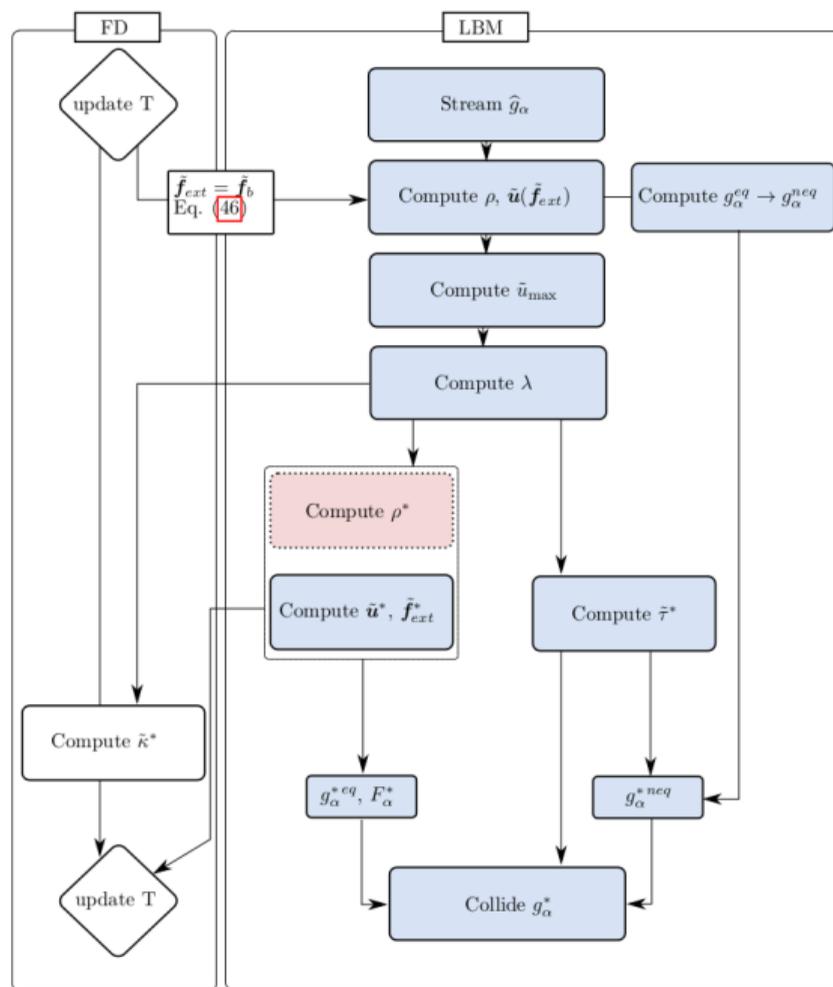
Diagram of the Adaptive Time-Stepping (ATS) algorithm

To avoid abrupt changes at the beginning of the simulation:

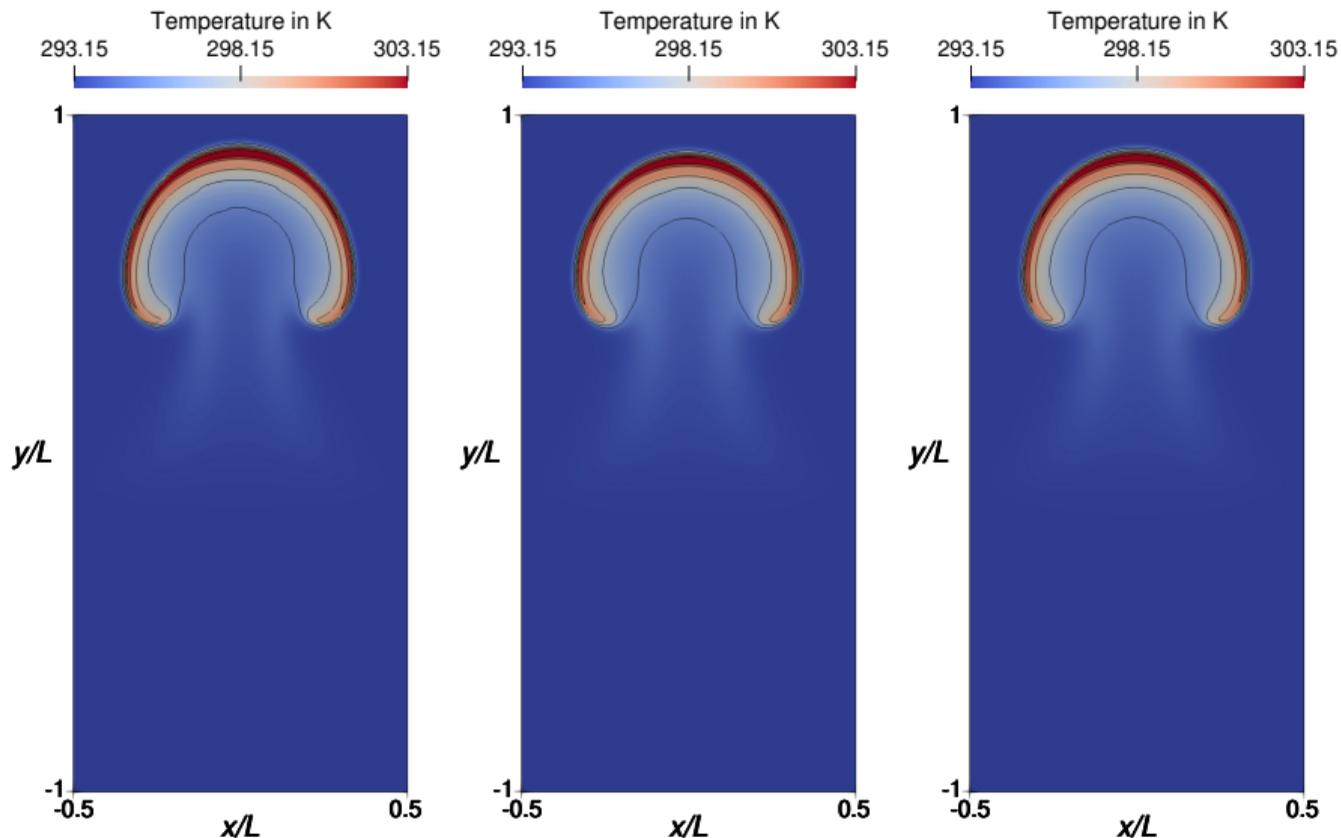
$$c_0^* = \max\left(\frac{u_{\max}}{\text{Ma}_t}, c_0^{0.9}\right)$$

where $\text{Ma}_t = 0.15$.

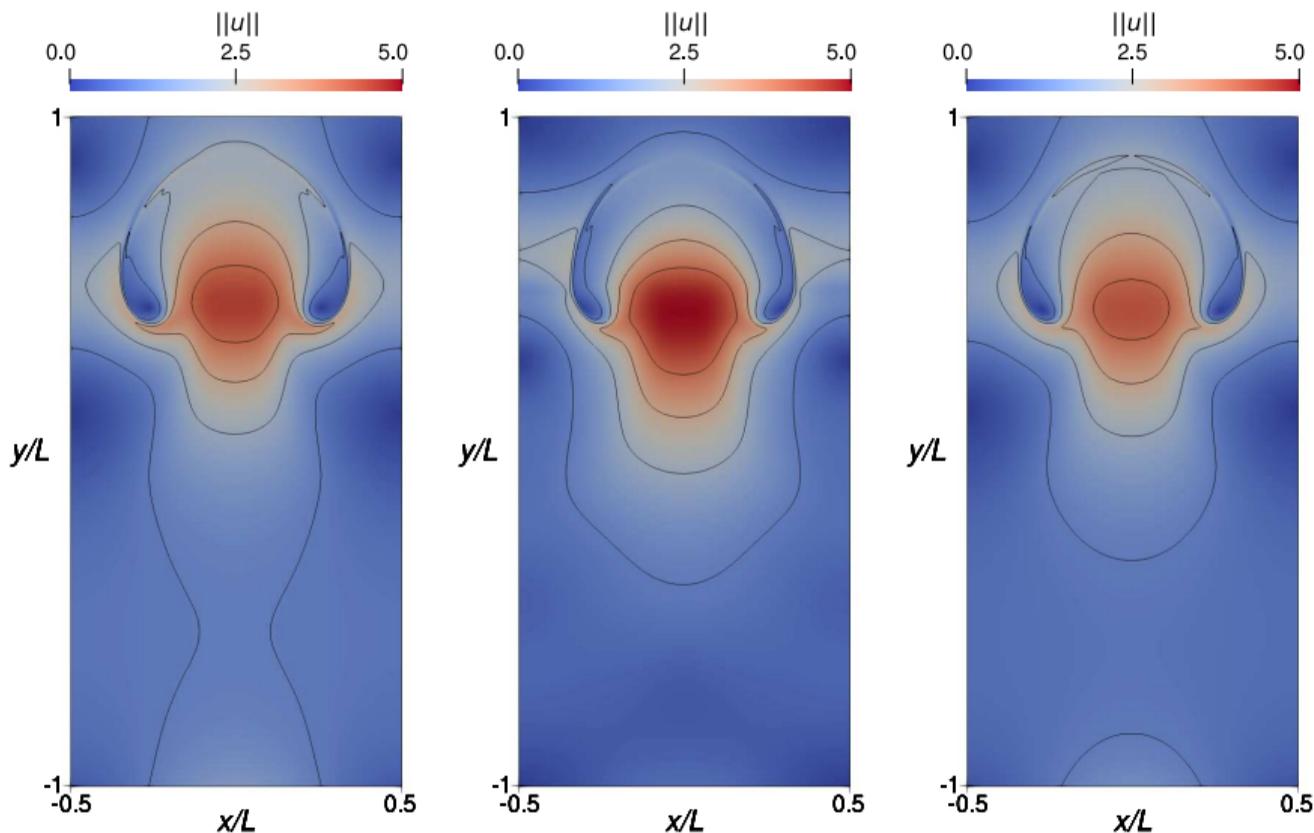
Δt is reevaluated every 10 iterations.



2D field: temperature [K] ($\Delta T = 10$) at 45.2 s

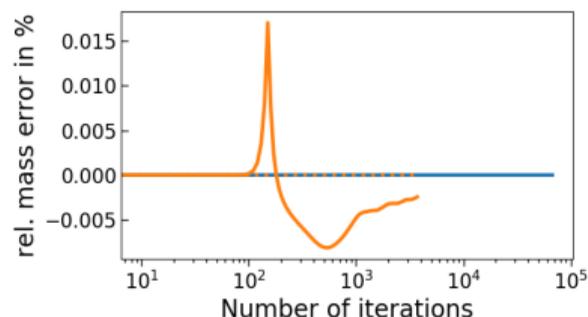
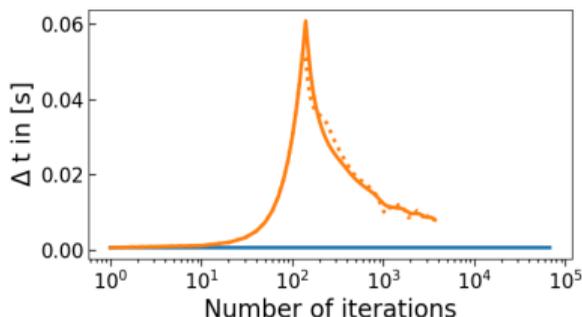
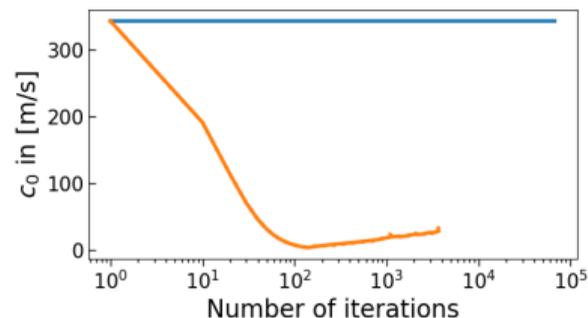
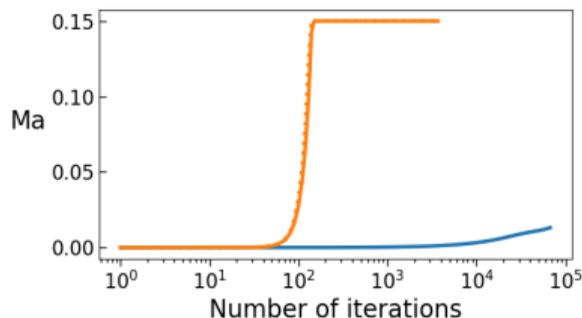


2D field: velocity norm [ms^{-1}] ($\Delta T = 10$) at 45.2 s



Rise of the thermal plume over a duration (physical time) of 45.3 s.

(—): constant time-step
(···): ATS
(—): ATS with correction
! Iterations in log-scale



- Qualitative agreement
- $\sim 20\times$ benefit for ATS
- ATS: oscillations at the end of the run (Δt plot)
- ATS with correction:
mass error
[-0.008%, 0.017%]

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Womersley flow: a pulsating 2D channel flow

The pressure gradient oscillates according to

$$\frac{\partial P}{\partial x} = A \cos(\omega t)$$

The problem has an exact solution in the laminar regime

$$u_x(y, t) = \Re \left\{ i \frac{A}{\rho \omega} \left(1 - \frac{\cos(\Lambda (\frac{2y}{L_y} - 1))}{\cos(\Lambda)} \right) e^{i \omega t} \right\}$$
$$u_y = 0$$

with $\Lambda^2 = -i \alpha^2$ and $\alpha^2 = \frac{L_y^2 \omega}{4 \nu}$.

Here $\alpha = 2.59$, $Re = 100$ and $\tilde{T} = 5000$
to mimic real life flow phenomena that can be encountered in the smaller arteries of the human body.

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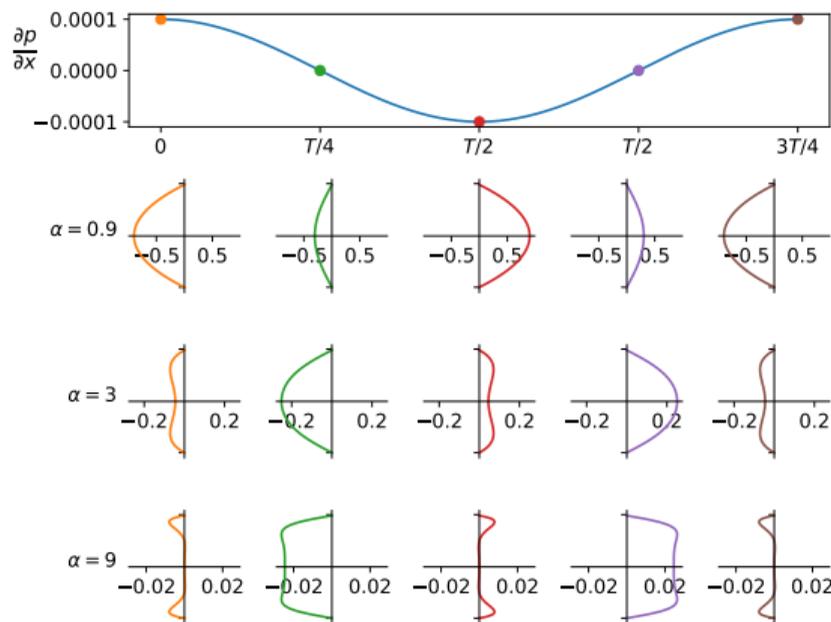
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Temporal evolution of the pressure gradient (top) and the corresponding velocity profile $u_x(y, t)/U_0$ for different values of the Womersley number α .

Womersley flow: body force vs inlet/outlet boundary conditions

The pressure gradient can either be established by two strategies:

1. an external **body force** to every nodes of the fluid

$$\mathbf{f}_{ext}(t) = -A \cos(\omega t) \mathbf{e}_x$$

2. the **inlet/outlet** density boundary conditions

$$\rho_{in}(t) = \rho_{ref} - \frac{AL_x}{2c_0^2(t)} \cos(\omega t) \quad \text{and} \quad \rho_{out}(t) = \rho_{ref} + \frac{AL_x}{2c_0^2(t)} \cos(\omega t).$$

This induces the pressure gradient through the momentum equation. It is expected that the continuity of the pressure forces will have an impact.

Womersley flow: probe at the center of the oscillating channel

(—): constant time-step

(-.-.-): ATS

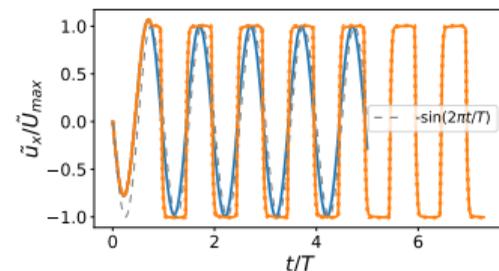
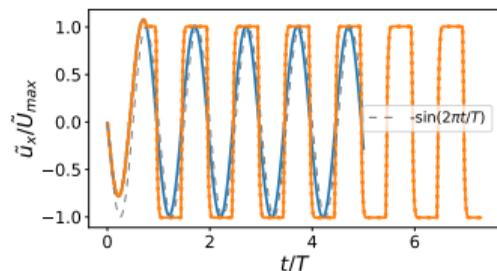
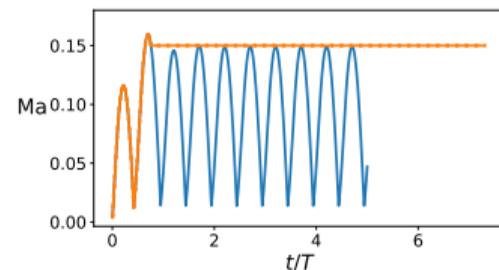
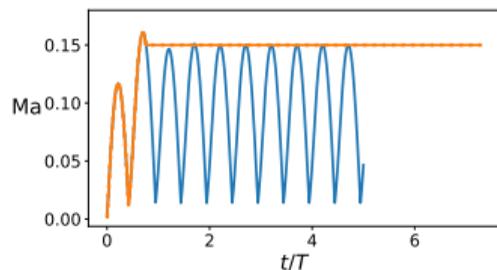
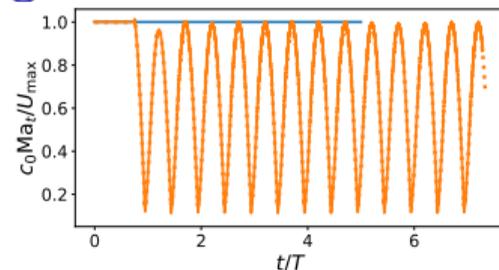
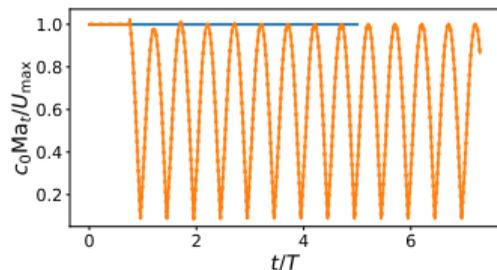
(—): ATS with correction

$Ma_t = 0.15$

ATS is activated at

$\tilde{t} = t/T = 0.75$.

- No difference between forcing strategies or correction in ATS
- Maximum theoretical gain:
 $\pi/2 \approx 1.57$
- ATS speedup $\times 1.53$



Left: body-force stirring

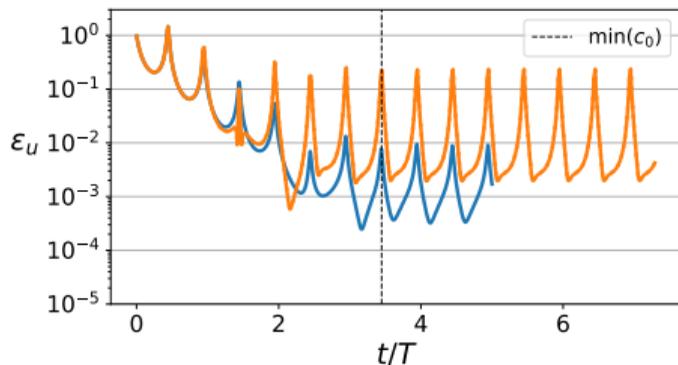
Right: inlet/outlet conditions

Womersley flow: velocity profile at the center of the oscillating channel

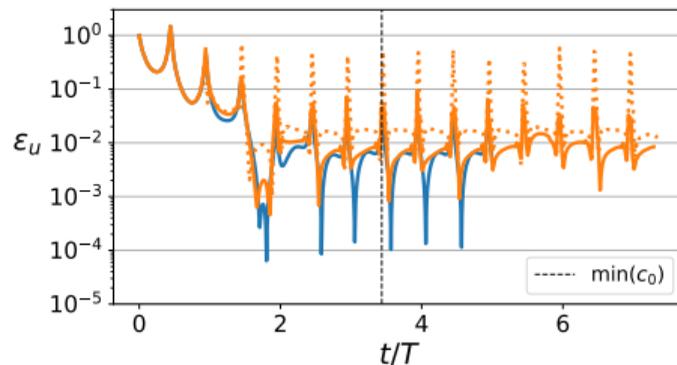
- (—): constant time-step
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$$\varepsilon_u(t) = \frac{\|u - \bar{u}\|_{L_2}}{\|\bar{u}\|_{L_2}} = \sqrt{\frac{\sum_y \left(u_x\left(\frac{L_x}{2}, y, t\right) - \bar{u}_x(y, t) \right)^2}{\sum_y \bar{u}_x(y, t)^2}}$$

where \bar{u}_x is the analytical solution.



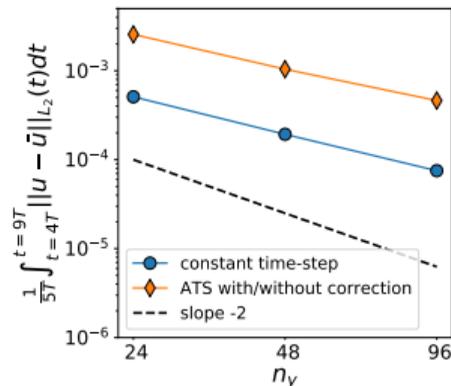
Left: body-force stirring



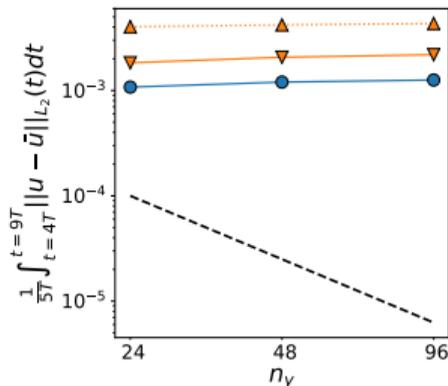
Right: inlet/outlet conditions

- Error peak when $\bar{u}_x \rightarrow 0$, $\varepsilon_u \rightarrow \infty$ (flow reversal).
- Body-force ATS errors: one order of magnitude higher wrt constant Δt .
- Inlet/outlet ATS with correction error: same order of magnitude wrt constant Δt .

Womersley flow: convergence rate of the ATS algorithms



Left: body-force stirring

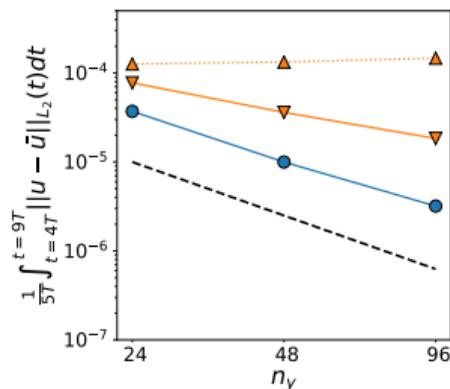
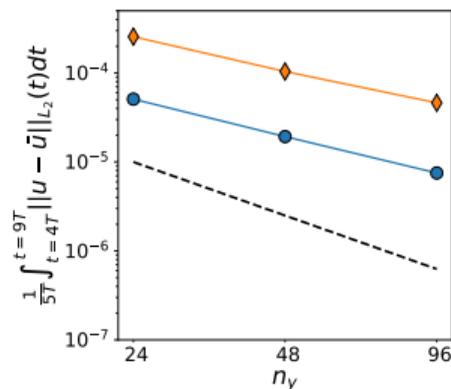
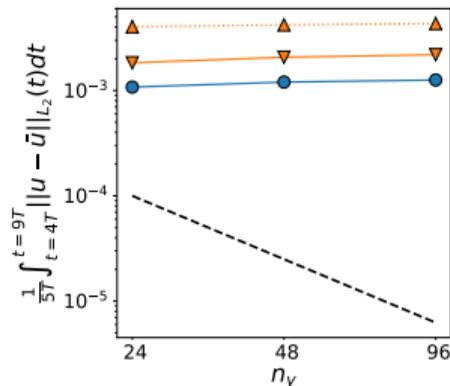
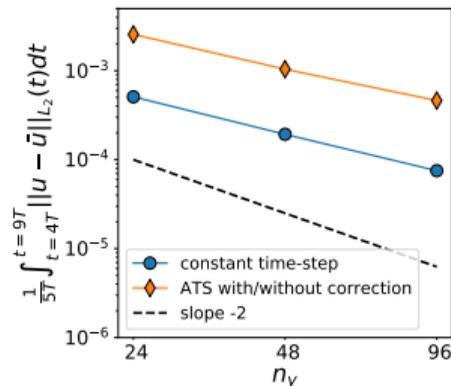


Right: inlet/outlet conditions

At $Ma_t = 0.15$

- Body-force ATS no influence of the correction
 - Inlet/outlet ATS error independent of the resolution
- ↪ Phase shift due to compressibility effects *wrt* incompressible solution

Womersley flow: convergence rate of the ATS algorithms



Left: body-force stirring

Right: inlet/outlet conditions

At $Ma_t = 0.15$

- Body-force ATS no influence of the correction
- Inlet/outlet ATS error independent of the resolution
- ↳ Phase shift due to compressibility effects *wrt* incompressible solution

At $Ma_t = 0.015$

- Inlet/outlet ATS 2nd-order convergence
- Inlet/outlet ATS with correction improves the accuracy

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 - Natural convection in thermal flow
 - Womersley flow
 - Channel entrance flow**
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Channel entrance flow

- Increase the previous channel length by $\times 20$
- Dirichlet pressure boundary condition at the outlet
- Dirichlet velocity boundary condition at the inlet
- Initial fluid at rest and initial ramp for the velocity at the inlet:

$$U_{\text{in}}(t) = \sin\left(\frac{\pi}{2} \frac{t}{T_{\text{ramp}}}\right) U_{\text{bulk}} \quad \text{for } t \leq T_{\text{ramp}}$$

$$U_{\text{in}}(t) = U_{\text{bulk}} \quad \text{for } t > T_{\text{ramp}}$$

with $U_{\text{bulk}} = 0.75U_0$ and $\tilde{T}_{\text{ramp}} = 10000 = \frac{1}{5}\tilde{T}_{\text{tot}}$.

- ATS is activated when $\text{Ma} > 0.03$

Probe at $x = 0.8L_x$,

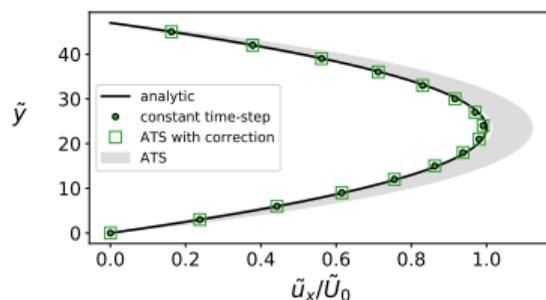
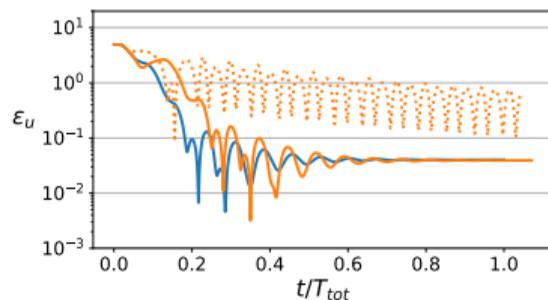
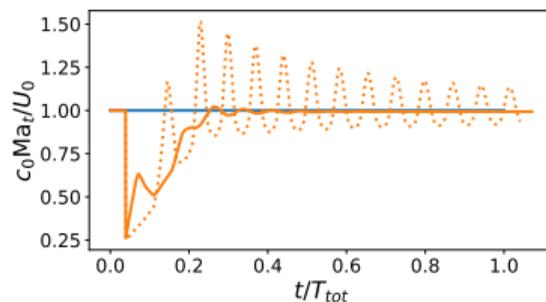
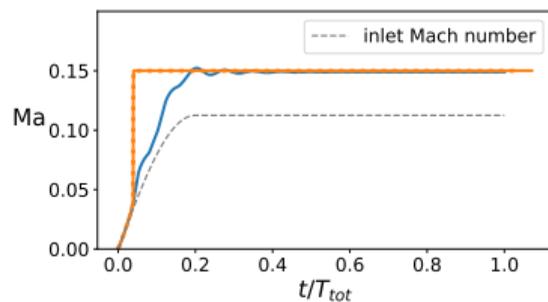
where the flow reaches a Poiseuille parabolic velocity profile ($U_{\text{max}} = U_0$) when $t \gg T_{\text{ramp}}$

Channel entrance flow: time evolution

(—): constant time-step

(····): ATS

(—): ATS with correction



ATS *without* correction: speed of sound overshoots and oscillates

There is no free-lunch

- ATS does not always improve the (initial) convergence.
- When $c_0 \searrow$, pressure waves take longer to dissipate.

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Conclusion

Adaptative time-stepping (ATS) in LB method

ATS introduces an error in either mass or momentum conservation:

- preserving the continuity of the density field instead of the pressure force (per unit mass)
- preserving the continuity of the pressure force (per unit mass) instead of the density field

Advantages of the ATS

- No prior knowledge about the maximum velocity is required
- Speed up can be considerable by optimally adapting on the flow dynamics:
e.g. biological flows, transient thermal flows and oscillating flows in general, where the maximum velocity undergoes large variations.

We show that

ATS does not always improve the (initial) convergence.

ATS with *pressure correction* performs better:

- natural convection
- channel flows with inlet/outlet boundary conditions

Conclusion

Defining the relative change of the time-step $\varepsilon = \frac{\Delta t^* - \Delta t}{\Delta t}$

- Continuity of the density field ($\rho^* = \rho$) yields a relative error on the pressure force

$$\left(\frac{p^* - p}{p}\right) \sim \left(\frac{c_0^{*2} - c_0^2}{c_0^2}\right) \sim \varepsilon.$$

- Continuity of the pressure force (per unit mass) yields a relative error on the density field

$$\left(\frac{\rho^* - \rho}{\rho}\right) \sim \varepsilon \frac{\rho - \rho_{\text{ref}}}{\rho} \sim \varepsilon \text{Ma}^2,$$

↪ This provides a plausible justification for the advantage of considering the continuity of the pressure force (per unit mass) in the ATS.

THANK YOU FOR YOUR ATTENTION

Cryospray ANR - open Ph.D. position

LATTICE BOLTZMANN SIMULATIONS OF THE DESTABILIZATION AND FRAGMENTATION OF A LIQUID INTO DROPLETS BY A FAST GAS STREAM



Numerical simulation of a kerosene flow destabilized by a crossflow of air

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